

Integral Users Group - MOC

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Flight Control Team



- Richard Southworth (Ops Manager, System, also 50% CHEOPS)
- Jutta Huebner (Ops Engineer, P/L, maternity leave until June 2019)
- Bruno Gandolfo (Ops Analyst, planning + Ops DB)
- Dave Salt (Ops Engineer, AOCS, also 50% XMM)
- Liviu Toma (Ops Engineer, AOCS, also 50% XMM)
- Stefano De Padova (Ops Engineer, MCS + OBDH)
- Timothy Finn (Ops Engineer, P/L, also 50% XMM)
- Norbert Pfeil (Ops Engineer, EPS + TCS + MCS, also 50% XMM)
- Thomas Godard (Ops Engineer, P/L + automation, also 50% XMM)

Temporary support:

- Jim Martin (my B/U during CHEOPS activities @ 20% in 2019)
- Frederic Schmidt (B/U for Jutta @ 30% until end 2019)
- Arnfried Magunia (B/U for Jutta @20% in 2019)





European Space Agency











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MOC



- No major activities currently
- Simulator ported to LINUX (old HW was bought in 1998!)
- Testing VIL2 (INTA) again, hopefully better performance than WHM
- KIR availability, reliability and coverage are very good







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1+1

SPACON Merger with Gaia



- Since 11/4/2018
- No significant issues for Integral
- Mission performance unaffected
 - Highly automated and Robust operational concept before the merger
 - Realistic planning at MOC and SOC
- All Software tools in place
 - Deconflicting Tool prevents planning of simultaneous events for all 3 missions
 - System Monitoring Tool detects critical problems when SPACON is busy elsewhere





Orbital Evolution

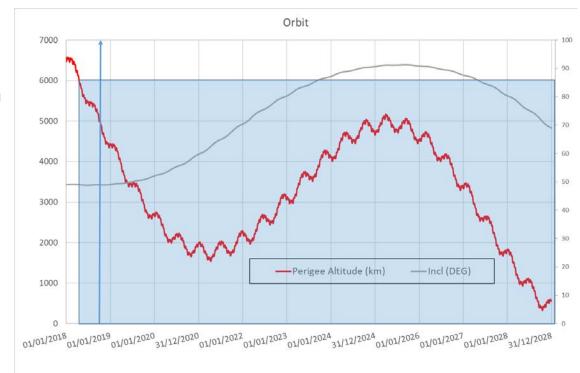


Perigee below 6000km until end of mission

- Faster rate of SA degradation
- Increased earth albedo exposure @ perigee
- Other consequences for components?

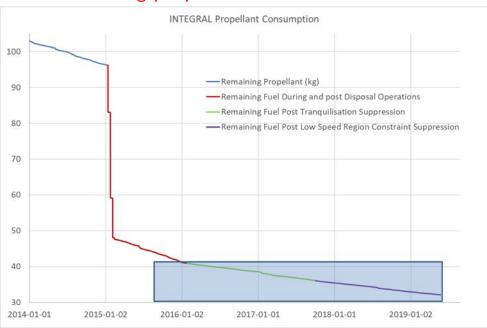
Increasing inclination good for station visibility

Re-entry in early 2029



Propellant Usage

Disposal manoeuvres in 2015 used half remaining propellant





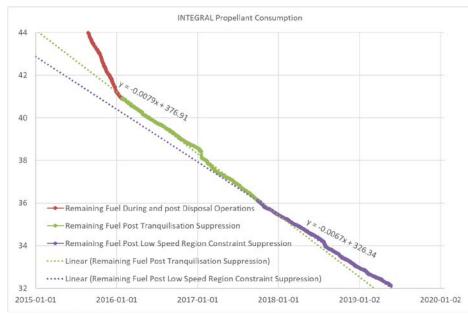


Stopping thruster tranquilisation post wheel bias:

20 to ~8g/day (attitude stability OK)

Removal of Reaction Wheels low speed constraint:

9g to ~7g/day (attitude stability + wheel health OK)



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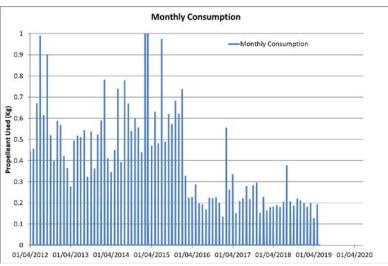






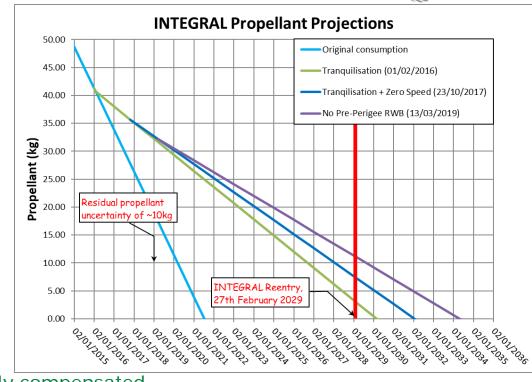
Propellant Usage and Predictions





Remaining 23/5/2019: 32.1kg

- Usage 2014: 6.7kg!!
- Usage 2017: 3.2kg
- Usage 2018: 2.5kg
- Expected 2019 onwards: ~2.2kg
- Effects of disposal manoeuvres fully compensated
- => Propellant will probably not be exhausted before re-entry



Array Degradation



From IUG 19

Action 19–3 on RS Due: end 2017

Prepare Technical Note on options for Eclipse Entry handling when array currents approach critical threshold.

2 issues:

- Power Budget evolution
- Autonomous reconfiguration ECL(s/e) at eclipse entry may occur in sunlight at high pitch angles => unplanned instrument switch-off

Draft TN available, clear strategy proposal for both issues

Minor outstanding detail re. testing

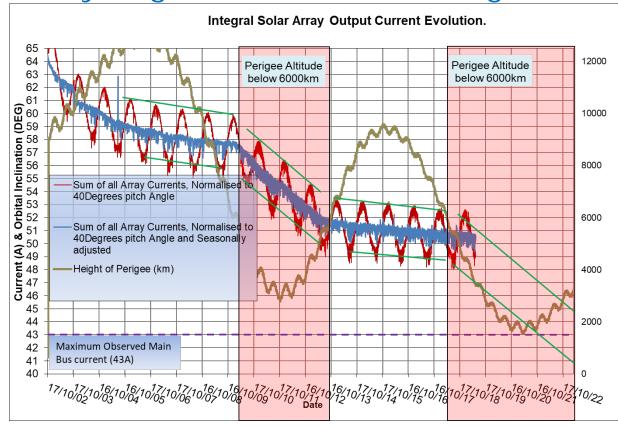
Exact timing is tbd, depends on further evolution







Array Degradation – Power Budget





EPS limits battery charge in case of excess demand

- Flexible margin of about 6A
- Occasional reduction in charge rate from 2021 (probably later)
- Mitigated by:
 - Long recharge time
 - Low Depth of Discharge

Limited discharge in sunlight allowable – **we know the** batteries are healthy

No power constraints expected before end 2022 (probably later)

Constraint only when discharge in sunlight is significant

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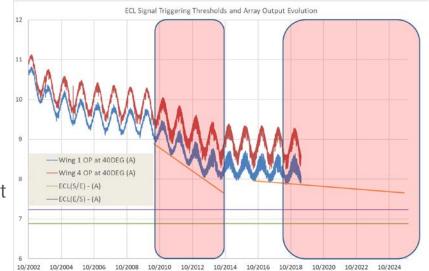
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Array Degradation – ECL (s/e)

The ECL Issue

- Autonomous reconfiguration at eclipse entry
- Based on array output I (Fixed threshold 6.9A)
- Reconfiguration for power (safety) reasons
- Approaching threshold at max pitch angle in sunlight
- Powers off PLM units unexpectedly long recovery!
- Not before 2020 probably later



The proposed solution

- Limit pitch angle to in eclipse season only, disable ECL(s/e) outside eclipse season,
 - Initially limitation to 35DEG is sufficient, later 30DEG
 - Ensure safety with OBM entry to re-enable ECL(s/e) signal at anomalously low array output (5A tbc)
 - Temporary loss of part of celestial sphere visibility for about 90 days / year

Start depends on observed degradation – lower than expected

Study of observed and predicted array degradation initiated ESA UNCLASSIFIED - For Official Use

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AOB



- New simulator available, based on LINUX, replaces VMS hardware
- Testing of INTA stations will start in summer as back-up to Kiruna
- Near real time Time Correlation delivery to ISDC started





















